College of Engineering Department of Mechanical & Industrial Engineering College of Engineering School of Electrical Engineering & Computer Science

To Predict ► To Design ► To Perform

ME, ECE Capstone Design Programs

Team 21 – SAE Aero Design Micro Class

Joe Bollich, Sterling Foley, Spencer Hancock, Khoa Nguyen, David Wale, Dylan Zeringue Pilot: Drew Rousseau

OBJECTIVE STATEMENT

The goal of this project is to design a competition winning RC aircraft under the constraints dictated by the 2018 SAE Aero Design Rules and Regulations. The team is to minimize aircraft weight, maximize payload, and reduce assembly time through utilization of the design process.

BACKGROUND INFORMATION

- **Competition Details**
- Van Nuys, CA

180°

turn

Launch

- 200 ft-

(Landing)

• April 6 – 8, 2018

400 ft

400 f



- Electric Motor/Lithium Polymer Battery
 Aircraft, payload, and tools stored in
- container

Competition Imposed Constraints

Payload: 2" SCH 40 PVC Pipe

Flight circuit (left)

ENGINEERING SPECIFICATIONS

180

turn

Measurable Specification Value		Description	
Max Total Weight	10 lbs.	Weight of aircraft, payload, container, and assembly tools	
Container Dimensions	13.875" x 12.125" x 3.625" t = 0.125"	Dimensions of cardboard carrying container	
Container Weight	1.5 lbs.	Weight of empty cardboard carrying case	
Payload Dimensions	OD = 2.375", ID = 2"	Outer/Inner diameter of PVC pipe	
Payload Specific Gravity	0.68 lbm./ft.	Density of PVC pipe per linear foot	
Battery Size	2-cell 325 mAh	Optimized battery size	
Assembly Time	1:45 min	Aircraft assembly time	
Flight Time	60 s	Maximum allowed time of flight	
Project Start Resea	rch Configuration De	uselage esign and System	
Aug. 26, 2017 Sep. 2	017 Sep. 2017 Se	ep. 2017 Oct. 2017	

COMPETITION RESULTS

Category	Score	Place	
Overall Score	92.3891	4 th	
Payload Fraction	0.69	3 rd	
Flight Score	29.5852	5 th	
Design Report	31.8640	6 th	
Technical Presentation	37.9400	9 th	

ASSEMBLED DESIGN





Jack	AT ACE COL
Rettig	

TESTING				
Test	Expected Result	Measured Result		
Static Thrust Test	> 140 grams	430 grams		
Wing Break Test	1.48 lbs.	7.11 lbs.		
Wing Deflection Test	1.7"	1.5"		
Battery Life Test	> 30%	58% Remaining		
Servo Test	4.16 ozin.	10.86 oz./ 60°		

SAFETY

- Red arming plug to arm and disarm system safely
- Programmed throttle failsafe to automatically shut system down if signal is lost
- Lithium Polymer battery charging bag to protect from battery fires
- Painted propeller tips
- Warming up and wearing a hard hat during launch

MATERIALS AND MANUFACTURING

Component		Material/Manufacturing	
Wing/Tail Ribs		Balsa Wood – laser cut from balsa wood sheets	
Wing Joints		Carbon Fiber – rods purchased and cut with bandsaw	
Wing Spars		Balsa Dowels – ordered and cut to length with bandsaw	
Wing Brack	et	Balsa Wood – pieces laser cut individually; glued together; sanded to shape	
Central Rods		Carbon Fiber Tubing – premade (ordered); cut to length with bandsaw	
Wing Bracket/Rib Reinforcement		Carbon Fiber Tubing – premade (ordered); cut to length with bandsaw	
Wing/Tail Skin		Coverite Microlite Film – cut to shape; ironed onto balsa wood ribs	
Purchasing Manufacturing Component Flight Testing Competition			
Dec. 2017 Jan./Feb. 20	018 Feb. 201	18 Mar. 2018 April 4-6, 2018	

Sponsors: Dr. K. Gonthier, LaSPACE, Mr. Jack Rettig, LSU MIE/EE Dept.

Advisers: Jack Hawkins, Mark Rabalais, Drew Rousseau